Fully Kinetic Simulations of Ion Beam Neutralization

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1. Introduction

Ion beam emission/neutralization is one of the most fundamental problems in spacecraft plasma interactions and electric propulsion. It is well known that, in order to transmit a current exceeding the space charge limit from spacecraft to the ambient, the beam must be adequately neutralized or the transmission would be blocked by virtual anode formation in the beam\(^1\) and spacecraft charging. Hence, the operation of an electric thruster or any other large current ion emitting source from spacecraft requires a neutralizer to neutralize the ion beam. In such systems, the ions are typically emitted as a cold beam while the electrons are typically emitted as stationary thermal electrons from the neutralizer. The emission is such that \(v_{te} >> v_{beam} >> v_{ti}\), where \(v_{te}\), \(v_{beam}\), \(v_{ti}\) are the electron thermal velocity, beam drifting velocity, and ion thermal velocity, respectively, and the electron current emitted equals the ion current emitted. One notes that the plasma emitted is strongly non-neutral near the source regardless of the neutralizer location or the emitting surface area. Even for a hypothetical situation where the electrons and ions were emitted from exactly the same location and same surface area, the initial beam would still be strongly non-neutral due to the difference in electron and ion emitting velocity.

The ion beam neutralization process not only is an interesting physics problem but also has important practical implications. For instance, such knowledge is important in the neutralization design for electric thruster clusters. It is obviously also of critical importance in any modeling studies involving plasma emission.

Ion beam neutralization is one of the first problems studied during electric propulsion development. Although ion beam neutralization is readily achieved in experiments, the understanding of the underlying physical process remains at a rather primitive level. No theoretical or simulation models have convincingly explained the detailed neutralization mechanism. Earlier theoretical and simulations models have considered the neutralization of infinitively large uniform ion beam\(^2-6\). More recent models have considered a more realistic setting for finite size ion beam emission\(^7-9\). These previous studies suggest that wave-particle interaction and plasma instability may be the driving neutralization mechanism. However, no conclusions have been reached. Part of the reason for the lack of a good understanding of the neutralization process is because particle simulation of beam neutralization is an extremely challenge problem due to computational constraints. This is because, in order to simulate the physics correctly, such simulations must be carried out using the realistic ion to electron mass ratio so the correct mesothermal velocity order for ions and electrons, \(v_{te} >> v_{beam} >> v_{ti}\) can be maintained. Additionally one must also use a very large simulation domain in order to minimize the effects of the simulation domain boundary.

This paper presents a fully kinetic simulation of ion beam neutralization and plasma beam propagation. The focus is on the physics of electron-ion coupling and the resulting propagation of the mesothermal plasma. Section
2 presents the simulation model. Section 3 discusses the simulation result. Section 4 contains a summary and conclusions.

2. Simulation Model

The ion beam neutralization process involves the following aspects: initial mixing of electrons and ions, electron-ion coupling, and beam propagation. The initial electron-ion mixing, to a large extent, is determined by device design and hence, the mixing process varies for different systems. In this paper, we will focus on the electron-ion coupling and beam propagation aspects.

The problem is studied using full particle PIC simulation. In this model, both the electrons and ions are modeled as macro-particles. The particle dynamics, space charge, and electric field are solved self-consistently. In order to reduce the computation, the 3-D PIC code is applied to a 2-D configuration. The simulation setup is shown in Figure 1.

We consider that the electrons and ions are emitted from the same surface area but with different velocity distribution functions. At every time step, Macro-particles representing the ions are emitted into the simulation domain as a drifting Maxwellian distribution and those representing the electrons as a stationary Maxwellian distribution. In order to maintain the realistic relative velocity ratio between the beam velocity, and electron and ion thermal velocities in the simulation, the simulations are performed using a realistic mass ratio of $m_i/m_e=1836$. Comparing to $V_{te}$, $V_{beam}$ and $V_{ti}$ are $V_{beam}=0.1 V_{te}$ and $V_{ti} = 0.0023 V_{te}$, respectively. These relative values are similar to typical ion thruster parameters.

The emitted electron and ion currents are kept the same. For a cold beam ions and thermal electrons, the electron and ion current density at the emitting surface are $J_{eo}=n_{eo}<V_{te}>$ and $J_{io}=n_{io}V_{beam}$, respectively, where $n_{eo}$ and $n_{io}$ denote the electron and ion density outside the emitting surface, respectively. For the $V_{te}$ and $V_{beam}$ considered here, $n_{eo} ~ 0.2n_{io}$. Hence, if the electrons and ions were uncoupled, such an emission would lead to a very non-neutral beam, as illustrated in Figure 2.

In the simulation, the cell size equals the Debye length calculated using $n_{io}$ and the electron temperature $T_e$ at the emitting surface. We consider a spacecraft with size 50X50. The beam emission width is $R_T=20$. The simulation domain is taken to be 600X400, or $30R_TX20R_T$. The potential at spacecraft body is fixed and while the potential at domain boundary is floating. The number of macro-particles near the emitting source is $\sim850/cell$ for each population and the total number of macro-particles used at end of run is typically around 7 million. Simulations were run using a time step resolution of $dt\approx0.1$, where $dt$ and $\omega_{pe}$ denote the time step and the electron plasma frequency, respectively.

3. Results and Discussions
Typical simulation results are presented in Figs. 3 through 8. Fig.3 shows potential contour at \( t_{\omega_{pe}}=1600 \) (\( t_{\omega_{pi}}=37.3 \) where \( \omega_{pi} \) denotes the ion plasma frequency ). Fig.4 shows electron and ion positions, electron density contour, ion density contour, and total charge density contour at \( t_{\omega_{pe}}=1600 \) (\( t_{\omega_{pi}}=37.3 \)). These results show that, while the beam is strongly non-neutral near the emitting source, the electron-ion coupling occurs immediately at the downstream of the emitting source and a quasi-neutral plasma beam quickly forms. The thermal electrons follow the motion of the cold beam ions, and the electron density closely matches the ion density inside the beam. For this particular case, the potential inside the beam at the downstream of the beam exit surface is only a few Te.

To investigate the process of electron and ion coupling, Figs. 5 through 7 show the time evolution of the phase plots, potential profiles, and electron and ion density profiles along the beam direction. In these plots, we compare the snapshots taken at \( t_{\omega_{pe}}=40 \) (\( t_{\omega_{pi}}=0.93 \)) with that at \( t_{\omega_{pe}}=1600 \) (\( t_{\omega_{pi}}=37.3 \)). The initial electron expansion along the beam direction follows the same physical process studied in 1-D expansion of a mesothermal plasma into vacuum. It is well understood that such expansion establishes an ion-acoustic like beam front propagation. As the electron thermal velocity is much larger than the ion beam velocity, the region behind the beam front will have a slightly positive potential with respect to the ambient. Hence, the region between the beam source and the beam front gradually traps the electrons. It is the interaction between the trapped electrons and the potential well that leads to electron-ion coupling and beam neutralization. Further frequency and wave number spectrum analysis (not shown here) also show that no beam plasma instabilities were present. We also performed the linear dispersion analysis using the plasma parameters such as the electron and ion velocities and densities observed in the potential well. However, the obtained

Figure 2: Illustration of a hypothetical non-neutral beam generated by the emission of cold beam ions and thermal electrons. a) electron (blue) and ion (red) positions; b) total charge density contour.

Figure 3: Simulation results: potential contour at \( t_{\omega_{pe}}=1600 \) (\( t_{\omega_{pi}}=37.3 \))
The growth rate of the beam instability is too small to grow. Therefore, in the current case, we find that ion beam neutralization is not through plasma micro-instability, as previous studies suggested. As the beam front propagates forward, the electrons and ions develop a similar density profile along the beam direction, as shown in Fig. 8.

Once the quasi-neutral beam is established, an expansion wave is generated outside the beam (Fig. 4c and 4d). The expansion in the transverse direction is similar to that associated with the self-similar expansion of a mesothermal plasma into vacuum.

4. Summary and Conclusions

In summary, we have developed a full particle PIC simulation model to simulate the ion beam neutralization process. We find that beam neutralization and propagation are two closely
coupled processes. The initial expansion of thermal electrons over cold beam ions establishes ion-acoustic-like beam front propagation. Subsequently, the emitted electrons are trapped in the region between the forward propagating beam front and the emitting source. Electron-ion coupling is achieved through the interactions between the trapped electrons and the potential well along the beam direction. Beam neutralization is not through plasma instabilities as previous studies suggested. Self-similar expansion of ion acoustic waves similar to that associated with plasma expansion into vacuum also occurs in the transverse direction outside the beam. Because of electron trapping in the beam direction and the interactions between the trapped electrons and the electric field, the electron

Figure 6: Potential profiles along the beam direction a) $t_{pe}=40 \ (t_{pi}=0.93)$ b) $t_{pe}=1600 \ (t_{pi}=37.3)$

Figure 7: Total charge density profile along the center axis a) $t_{pe}=40 \ (t_{pi}=0.93)$ b) $t_{pe}=1600 \ (t_{pi}=37.3)$

Figure 8: Ion density profile (a) and electron density profile (b) along the center axis at $t_{pe}=1600 \ (t_{pi}=37.3)$
distribution is highly non-Maxwellian along the beam direction. Hence, the commonly used Boltzmann assumption for electron density in spacecraft plasma interaction models in general is not valid for interactions concerning plasma beam emission.

**Reference**


